# 1993 NASA AEROSPACE BATTERY WORKSHOP

# ENVIRONMENTAL SURVEY (MESUR) PATHFINDER BATTERY STUDY FOR THE MARS

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## MESUR PATHFINDER - BATTERY WORKSHOP BATTERY STUDY OVERVIEW

- MESUR PATHFINDER INTRODUCTION
- POWER SUBSYSTEM CONCEPT
- BATTERY TECHNOLOGY SELECTION
- MISSION BATTERY PERFORMANCE
- CELL/BATTERY BASELINE DESIGN
  - CHARGE METHODOLOGY
- PROPOSED TESTING

### MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY Ag/Zn INTRODUCTION

PROVIDE AND STORE ELECTRICAL POWER FOR THE MESUR BATTERY MISSION PROFILE

LANDER DURING LAUNCH, CRUISE, ENTRY - DESCENT LANDING, AND MARS SURFACE OPERATION

TECHNOLOGY SELECTION

PERFORMANCE

SPECIFIC ENERGY DENSITY (0° C)

**VOLUMETRIC ENERGY DENSITY** 

RATE CAPABILITY

CYCLE LIFE

~50 CYCLES

~75 Wh/Kg ~158 Wh/L ~C RATE

FLIGHT HERITAGE (GENERIC)

SURVEYOR, MARINER, PIONEER, TITAN, VIKING

**CELL AND BATTERY DESIGN GENERIC** 

General Session

## MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY Ag/Zn HERITAGE

UNKNOWN	CLASSIFIED	YES	one			8	1.5	6 TO 20	50 (100)	100	12	,	0	SZHK30	6.48	097	3	1.6		1 WEBRIL	6 CELLOPHANE		1 VISKON
LARGE PROBE	PIONEER/VENUS	YES	19	13.4	33.3x19.3x14.4	\$	28.5	10 to 50	1 (25)	8	<b>1</b> 2	<u></u>	c	SZLR40	0.51	27.1	9	1.46		1 PELLON	6 CELLOPHANE		1 VISKON
MAR - 4333	LOCKHEED / SAT.	YES	16	62.7	50.8x34.3x20.3	300	24	9 TO 12	3200 (6700)	1 to 4	<b>o</b>	2	7	2599-3	2.951	1066	\$	1.04		1 PELLON	5 CELLOPHANE	1 PVA	1 VISKON
SAR -4265	APOLLO	YES	8	12.9	29.9x17.4x14.6	<b>\$</b>	8	25-35	6 (20)	80 (max)	12	7	1	1560-7	0.51	27.1	\$	1.9		1 PELLON	4 CELLOPHANE	1 PVA	1 VISKON
CELL	US ARMY	ΝΑ	one			2	5.1	10 TO 60	100 (200)	62.6	12	12	NA	BB-465/U	0.581	373	42	1.67		1 DYNEL	5 CELLOPHANE		1 DYNEL
MAR 4557-X	MESUR	YES	18	13.3	22.5x21x18.1	\$	7.7	-	8	80 (max)	15	2	<b>∞</b>	SZLR 40-3	0.677	278	\$	1.63		1 WEBRIL	<b>5 CELLOPHANE</b>		1 VISKON
ВАТТЕКУ	PROGRAM	FLIGHT EXPERIENCE	NUMBER OF CELLS	WEIGHT (Kg)	SIZE (cm)	CAPACITY (A-H)	VOLTAGE (NOMINAL)	RATE (AMPERES)	CYCLE LIFE (actual)	DEPTH-OF-DISCH. (%)	WET LIFE (MONTHS)	GROUND (MONTHS)	FLIGHT (MONTHS)	CELL PART No.	CELL WEIGHT (Kg)	CELL Vol. (cm^3)	ELECTRO. CONC. (%)	NEG. to POS. RATIO	SEPARATION SYSTEM	POS. ABSORBER*	MEMBRANE	MEMBRANE	NEG. ABSORBER

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### MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY BATTERY OVERVIEW

PERFORMANCE DRIVEN DESIGN

OPERATIONAL PARAMETERS WILL BE DETERMINED BY CELL AND BATTERY PERFORMANCE

DERIVED PERFORMANCE TARGETS

BATTERY VOLTAGE 22-36 VOLTS

LOAD 1-4 AMPERES

TWO MONTHS ACTIVE STORAGE 20°C (SHIP AND LAUNCH)

SEVEN MONTH CRUISE

30 CYCLES MARS SURFACE 20 Ah AT 10 TO -20°C

CAPACITY

40 AH AT RT AND BOL

20 Ah AT 0°C

VOLUME: 8.3 X 7.37 X 7.12 INCHES (ESTIMATE)

MASS: LESS THAN 14 KG

## MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY CELL DESIGN

## • CELL BASELINE DESIGN

ELECTRODE	POSITIVE	NEGATIVE
MATERIAL	SINTERED Ag 4	Zn OXIDE BLEND 5
SURFACE AREA	124.5	
THEORETICAL	66.5 AH	101.4 AH
INITIAL EXPECTED	54.5 Ah	
SEPARATOR SYSTEM	5/6 LAYERS - CELLOPHANE	OPHANE
ELECTROLYTE	40 % KOH, 85 ml	
DIMENSION	6.61 X 3.54 X 0.725 INCHES	INCHES
WEIGHT	445 grams DRY (APPX)	PPX)

### MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY BATTERY DESIGN

BATTERY BASELINE DESIGN

BASED ON CURRENT 20 CELL, 3 CONNECTORS, TI HOSING

**18 CELL CELL BATTERY** 

THREE CONNECTOR. MAIN POWER, INSTRUMENTATION, HEATER POWER

**DIMENSIONS** 8.30 X (8.87) X 7.12 INCHES

FOOTPRINT TBD (HOLE PATTERN) FLANGE 0.75 INCH

HEATER SET POINT APPX 80 F, POWER TBD, THERMOSTATS TBD

WEIGHT (ACTIVE BATTERY) ESTIMATES

18 CELLS	22.86	TOTAL = 29.25 lbs
HOUSING	3.22	13.28 KG
COMPOSITE	1.07	EXPECT WEIGHT GROWTH
CONNECTORS	0.28	
WIRING	0.40	
HEATER	0.03	
MISC	1.39	

### MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY DESIGN/PERFORMANCE ISSUES

DESIGN

DESIGN BASED ON EXISTING TECHNOLOGY

**CELL 5 OR 6 LAYERS OF SEPARATOR** 

18 CELL BATTERY (CONCEPT BASED ON 20 CELL BATTERY)

OPERATIONAL

CHARGE

**ENERGY BALANCE IS CRITICAL** 

20 Ah CHARGE IN 6 HOURS

CHARGE/DISCHARGE TEMPERATURE LIMITS (PREFER 20°C)

CHARGE METHODOLOGY (CP OR TAPER)

DISCHARGE

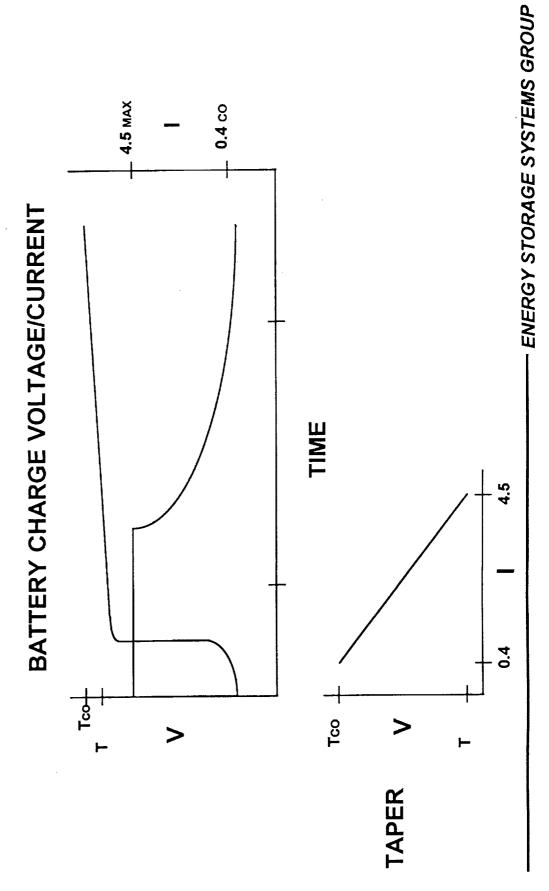
TEMPERATURE LIMIT FOR CAPACITY (NOMINAL 20Ah)

CRUISE

TEMPERATURE 0C PREFERRED WILL TEST AT 20°C

OCV OR 1.86 VOLT FLOAT

# MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY PROPOSED CHARGE METHODOLOGY



### **MESUR PATHFINDER - BATTERY WORKSHOP** CELL CHARACTERIZATION TESTING LANDER Ag/Zn BATTERY

CHARGE DEFINITION

1993 NASA Aerospace Battery Workshop

CONSTANT POTENTIAL

V = 1.91 TO 1.94, I = 4.5 MAX TO 0.4, TEMP., 23 TO 10C

VIRTUAL RESISTANCE

 $V_{vo} = 1.91 \text{ TO } 1.94, VT_{os} = 1.91, TEMP. = -10 TO 10C$ 

CHARGE/STAND CYCLE LIFE

8 MONTH STAND, 0 AND 25C, OCV AND 1.86V FLOAT

DISCHARGE CHARACTERISTIC

TEMP. -40, -20, -10, 0, 10C, RATES 1.5, 4.35, 12.5, 1.5 AMPS

SYSTEM PERFORMANCE

FLIGHT - PRELAUNCH, CRUISE/TCM, EDL, MARS SURFACE OPERATION

CYCLE LIFE INITIAL

CHARGE/DISCHARGE CYCLE AT 0C TO 1.22V, TAPER TBD

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## MESUR PATHFINDER - BATTERY WORKSHOP LANDER Ag/Zn BATTERY DESIGN OPTIONS DETERMINED BY TESTING

HARDWARE

**18 OR 17 CELL BATTERY** 

5 OR 6 LAYERS OF SEPARATOR

**OPERATION** 

CHARGE

LIMITED CHARGE MAY LEAD TO NEGATIVE ENERGY BALANCE

**DURATION OF CHARGE PERIOD AND CHARGE ACCEPTANCE** 

CHARGE TEMPERATURE COMPENSATION REQUIRED

CHARGE METHODOLOGY

CRUISE

TEMPERATURE, OCV OR 1.86 VOLT FLOAT

# **MESUR PATHFINDER - BATTERY WORKSHOP**

### SUMMARY

SELECTION OF Ag/Zn BATTERY BASED ON DERIVED MISSION REQUIREMENTS. •CELL AND BATTERY DESIGNS ARE CONSISTENT WITH FLIGHT REQUIREMENTS.

CHARGE/DISCHARGE TEMPERATURE COMPENSATION **•CELL TESTING WILL PROVIDE DATA ON** CHARGE METHODOLOGY **CRUISE REQUIREMENTS** CYCLE LIFE

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